

Available online at www.sciencedirect.com

Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

Computer methods in applied mechanics and engineering

www.elsevier.com/locate/cma

A discontinuous Galerkin method for two-temperature plasmas

Guang Lin, George Em Karniadakis *

Division of Applied Mathematics, Center for Fluid Mechanics, Brown University, Box 1966, 37 Manning Street, Providence, RI 02912, USA

Accepted 17 June 2005

Abstract

We develop a formulation for the single-fluid/two-temperature equations for simulating two-species, compressible, non-equilibrium plasma flows. The divergence-free condition of the magnetic field is enforced via the characteristic decomposition of an extended nine-wave system. The source terms are modified appropriately to improve energy and momentum conservation accuracy. A spectral/*hp* element algorithm is employed in the discretization combined with a discontinuous Galerkin formulation for the advective and diffusive contributions. The formulation is conservative, and monotonicity is enforced by appropriately lowering the spectral order around discontinuities. A new MHD flux introduced here is the MHD-HLLC (Harten–Lax–van Leer Contact wave) flux that preserves monotonicity and resolves contact discontinuities better. Exponential convergence is demonstrated for a magneto-hydrostatic problem. Two tests are presented using the new MHD-HLLC flux. Also, the differences between the single-temperature and the two-temperature models are presented for two-dimensional plasma flows around bluff bodies. © 2005 Elsevier B.V. All rights reserved.

Keywords: Viscous MHD; Spectral elements; Non-equilibrium plasma

1. Introduction

Plasmas can be modelled accurately using kinetic theory, especially partially ionized plasmas. However, this involves solutions of the seven-dimensional Boltzmann equation coupled with Maxwell's equations, which is prohibitively expensive. Particle-based methods, such as DSMC, are possible alternatives but for efficiency they need to be coupled with continuum fluid equations. Such hybrid methodologies have

* Corresponding author. Tel.: +1 401 863 1217; fax: +1 401 863 3369. *E-mail address:* gk@dam.brown.edu (G.E. Karniadakis).

^{0045-7825/\$ -} see front matter @ 2005 Elsevier B.V. All rights reserved. doi:10.1016/j.cma.2005.06.024

been used successfully recently in the simulation of ion thruster plumes, but hybrid kinetic-continuum methods are still under development [1,2], and open issues with the DSMC remain the treatment of electrons as well as the modelling of charged particle collisions.

Continuum-based, i.e., purely fluid approaches, have been successful in describing the macroscopic features of high density plasmas in many diverse applications [3–10]. They are derived from the Boltzmann equation by taking appropriate moments for each species. The standard mathematical description is that of single-fluid MHD with magnetic and gas dynamic viscous effects. However, a single-fluid MHD description has its limitations as it cannot account for local thermodynamic non-equilibrium effects and cannot consider non-neutral regions and sheath interactions. To this end, two-fluid plasma models [11] and corresponding solvers have been under development more recently [12]. They can overcome certain limitations of the single-fluid MHD model such as the Hall effect and diamagnetic terms, which model contributions to ion current and the finite Larmor radius of the plasma constituents. However, they still assume local thermodynamic equilibrium within each fluid. From the computational standpoint, the two-fluid model is much more complex system to solve, especially for large values of the Hall parameter, and approximate Riemann solvers are still under development [12].

In between the single-fluid and the two-fluid models for plasmas is the *two-temperature model*. It can account partially for the energy transfer between heavy species and electrons, and it is computationally more tractable. There can be many applications for which the electron temperature differs from the heavy particle temperature [13–15]. For example, experiments in [13] with boundary layers of pure NaK seeded argon showed that the electron temperature was considerably higher than the gas temperature. Moreover, the electrical conductivity and other transport coefficients in the conservation laws depend strongly on the electron temperature. This is also true for emerging applications in microfluidics, such as micro-pulsed plasma thrusters (micro-PPTs) where viscous effects dominate, and the flow may be in a transitional state.

The objective of the current work is to develop robust high-order algorithms for a single-fluid/two-temperature plasma extending our previous work on the standard MHD model [16]. The use of high-order accuracy addresses effectively the small-scale requirements of compressible MHD turbulence [17], as well as the extra resolution required in long-time integration [18]. In addition to our work, other recent efforts to develop effective high-order methods for plasma flows have been reported in [19–21].

Discontinuous Galerkin methods [22] address two of the main difficulties in employing *high-order discretization* for the solution of hyperbolic conservation laws:

- 1. maintaining monotonicity for non-smooth solutions, and
- 2. preserving conservativity.

In the MHD framework, such difficulties are compounded by the imposition of the *divergence-free* condition for the magnetic field, which results in a loss of the hyperbolicity of the ideal MHD equations. This condition has been dealt with by employing staggered grids in the work of Evans and Hawley [23], which was extended by Peterkin et al. [5]. However, such an approach cannot be easily incorporated in high-order discretizations. Alternative approaches include the operator-splitting algorithm proposed by Zachary et al. [4] and the development of extended Riemann solvers by Powell [6]; the latter is easily extended to multidimensions and also to high-order discretization. In some approaches, the divergence-free condition is not imposed directly during time-stepping but the initial conditions are projected in the divergence-free space. Assuming that the *flux* of divergence of the magnetic field satisfies a homogeneous discrete parabolic equation with homogeneous boundary conditions, then this will lead to zero discrete divergence at all times. However, in practice, discretization errors or other inconsistencies may trigger large divergence errors for such cases.

In this paper we extend the approximate Riemman solver of Powell [6] to a nine-wave system to account for the divergence-free condition and the extra electron energy equation. Compared to the Riemann solver

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

[6] and also our previous work in [16], here we modify the source terms to preserve conservativity and enhance accuracy. We employ a spectral/*hp* element discretization [24] based on tensor-product polymorphic elements for the mesh macro-skeleton. The spectral order of the Jacobi polynomials in the trial basis is variable in order to accommodate different solution requirements, i.e., low-order for shock capturing or high-order for boundary layer resolution.

To preserve pressure positivity, a MHD-HLLC flux [25] is implemented in the discontinuous Galerkin method for solving compressible plasma flows. In addition to capturing the effects of contact waves, the MHD-HLLC flux also resolves Alfven and slow waves better than the HLL (Harten–Lax–van Leer) flux and the Lax–Friedrichs flux.

As a model problem we use plasma flow past a cylinder. Depending on the specific conditions, we find that the electron temperature can be substantially different than the temperature of ions, and correspondingly this may affect the velocity field. To appreciate the differences we compare the two-temperature model with the standard MHD model under the same wall thermal condition.

The paper is organized as follows: In Section 2 we present the formulation and briefly summarize details of the implementation. In Section 3, we first test the convergence rate of the algorithm for an analytical problem. We then present tests using the new MHD-HLLC flux, and subsequently we simulate plasma flow past a cylinder in the subsonic and supersonic regimes. Finally, we conclude in Section 4 with a few remarks.

2. Two-temperature plasma equations

2.1. Governing equations

The non-dimensional governing equations for single-fluid/two-temperature plasma for compressible magneto-hydrodynamics (MHD) can be expressed in conservative form as (see derivation in Appendix B):

1. Mass conservation

$$\frac{\partial \rho}{\partial t} = -\nabla \cdot (\rho \mathbf{v}). \tag{1}$$

2. Momentum conservation

$$\frac{\partial(\rho \mathbf{v})}{\partial t} = -\nabla \cdot \left(\rho \mathbf{v} \mathbf{v}^{t} - \mathbf{B} \mathbf{B}^{t} + \left(p + \frac{1}{2} |\mathbf{B}|^{2}\right) I - \frac{1}{S_{vi}} \boldsymbol{\tau}_{i} - \frac{1}{S_{ve}} \boldsymbol{\tau}_{e}\right).$$
(2)

3. Magnetic field

$$\frac{\partial \mathbf{B}}{\partial t} = -\nabla \times \left(\mathbf{B} \times \mathbf{v} + \frac{1}{S_r} \nabla \times \mathbf{B} \right). \tag{3}$$

4. Total energy conservation

$$\frac{\partial E_{\text{tot}}}{\partial t} = -\nabla \cdot \left[(E_{\text{tot}} + p) \mathbf{v} + \left(\frac{1}{2} |\mathbf{B}|^2 \mathbf{I} - \mathbf{B} \mathbf{B}^{\text{t}} \right) \cdot \mathbf{v} - \frac{1}{S_{\text{vi}}} \mathbf{v} \cdot \tau_{\text{i}} - \frac{1}{S_{\text{ve}}} \mathbf{v} \cdot \tau_{\text{e}} \right. \\ \left. + \frac{1}{S_r} \left(\mathbf{B} \cdot \nabla \mathbf{B} - \nabla \left(\frac{1}{2} |\mathbf{B}|^2 \right) \right) - \frac{\gamma}{S_{\text{ve}} P r_{\text{e}}} \nabla T_{\text{e}} - \frac{\gamma}{S_{\text{vi}} P r_{\text{i}}} \nabla T_{\text{i}} \right].$$

$$(4)$$

5. Electron internal energy conservation

$$\frac{\partial \epsilon_{\rm e}}{\partial t} = -\nabla \cdot \left[(\epsilon_{\rm e} + p_{\rm e}) \mathbf{v} - \frac{\gamma}{S_{\rm ve} P r_{\rm e}} \nabla T_{\rm e} \right] + \mathbf{v} \cdot \nabla p_{\rm e} + \frac{1}{S_{\rm ve}} \tau_{\rm e} : \nabla \mathbf{v} + \frac{1}{S_r} (\nabla \times \mathbf{B}) \cdot (\nabla \times \mathbf{B}).$$
(5)

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

6. Magnetic flux constraint

$$\nabla \cdot \mathbf{B} = \mathbf{0}.$$

7. Ohm's law

$$\mathbf{E} = \eta \mathbf{J} - \mathbf{v} \times \mathbf{B}.\tag{7}$$

Here we define

$$E_{\text{tot}} = \frac{p}{(\gamma - 1)} + \frac{1}{2} (\rho \mathbf{v} \cdot \mathbf{v} + \mathbf{B} \cdot \mathbf{B}), \quad p = p_{\text{i}} + p_{\text{e}}, \quad \epsilon_{\text{e}} = \frac{P_{\text{e}}}{\gamma - 1}.$$
(8)

The stress tensor for ions and electrons is defined as

$$\boldsymbol{\tau}_{i} = (\partial_{j} \mathbf{v}_{ii} + \partial_{i} \mathbf{v}_{ij}) - \frac{2}{3} \nabla \cdot \mathbf{v}_{i} \delta_{ij}, \quad \boldsymbol{\tau}_{e} = (\partial_{j} \mathbf{v}_{ei} + \partial_{i} \mathbf{v}_{ej}) - \frac{2}{3} \nabla \cdot \mathbf{v}_{e} \delta_{ij}.$$
(9)

All other parameters are as defined in Table 1. The subscript 'i' denotes ions while the subscript 'e' denotes electrons.

The above Ohm's law is simplified from the generalized Ohm's law in the limit of *small Larmor radius* approximation. The generalized Ohm's law can be expressed as

$$\mathbf{E} = \eta \mathbf{j} + \mathbf{v} \times \mathbf{B} + \frac{\mathbf{j} \times \mathbf{B}}{n_{\rm e}} - \frac{\nabla p_{\rm e}}{n_{\rm e}}$$
(10)

Table 1			
Variables and parameters	used in the equations	of single-fluid/two-tempera	ture compressible MHD

Variable	Description
$\rho(\mathbf{x}, t) = \rho_i + \rho_e$ $\mathbf{v}(\mathbf{x}, t) = (u, v, w)(\mathbf{x}, t)$ $\mathbf{B}(\mathbf{x}, t) = (B_x, B_y, B_z)(\mathbf{x}, t)$	Single-fluid density Single-fluid velocity Magnetic fields
$E = \frac{p}{(v-1)} + \frac{1}{2}(\rho \mathbf{v} \cdot \mathbf{v} + \mathbf{B} \cdot \mathbf{B})$	Total energy
$p = p_{\rm i} + p_{\rm e}$	Total pressure
$\bar{p} = p + \frac{1}{2} \mathbf{B} \cdot \mathbf{B}$	Pressure plus magnetic pressure
$T_{\rm i} = \frac{p_{\rm i}}{\rho_{\rm i}}, T_{\rm e} = \frac{p_{\rm e}}{\rho_{\rm e}}$	Ion and electron non-dimensional temperature
$Pr_{i} = \frac{c_{p}\mu_{i}}{r_{e}}, Pr_{e} = \frac{c_{p}\mu_{e}}{r_{e}}$	Ion and electron Prandtl number
η κ_i κ_e	Magnetic resistivity
$\mu_{\rm i}, \mu_{\rm e}$	Ion and electron viscosity
$S_{\mathrm{vi}} = \frac{\rho_0 V_{\mathrm{A}} L_0}{\mu_{\mathrm{i}}}, S_{\mathrm{ve}} = \frac{\rho_0 V_{\mathrm{A}} L_0}{\mu_{\mathrm{e}}}$	Ion and electron viscous Lundquist number
$S_r = \frac{\mu_0 V_A L_0}{\eta}$	Resistive Lundquist number
c _p	Specific heat at constant pressure
$V_{\rm A}^2 = \frac{\mathbf{B} \cdot \mathbf{B}}{\mu_0 \rho}$	Alfven wave speed
$A = \sqrt{\frac{\mathcal{V}_{\rm A}^2}{\mathcal{V}_0^2}}$	Alfven number

4

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

and thus

$$\frac{\mathbf{j} \times \mathbf{B}/ne}{\mathbf{v} \times \mathbf{B}} \sim \frac{\nabla p_{\rm e}/ne}{\mathbf{v} \times \mathbf{B}} \sim \frac{r_{\rm Li}}{L},\tag{11}$$

where r_{Li} is the ion Larmor radius and L is the scale-length of the fluid motion. We conclude that the second and third terms on the right-hand side of Eq. (10) can be neglected if the ion Larmor radius is very small compared to the characteristic length scale of the fluid motion, i.e., $r_{\text{Li}}/L \ll 1$. Specifically, when we consider the length scale of the fluid motion to be very small and close to the ion Larmor radius, then we have to include the two additional terms in Eq. (10) and use the generalized Ohm's law instead.

Alternatively, in flux form, the above conservation equations can be expressed compactly as

$$\frac{\partial \mathbf{U}}{\partial t} = -\frac{\partial \mathbf{F}_{x}^{\text{Ideal}}}{\partial x} - \frac{\partial \mathbf{F}_{y}^{\text{Ideal}}}{\partial y} - \frac{\partial \mathbf{F}_{z}^{\text{Ideal}}}{\partial z} \tag{12}$$

$$+\frac{\partial \mathbf{F}_{x}^{\text{Visc}}}{\partial x}+\frac{\partial \mathbf{F}_{y}^{\text{Visc}}}{\partial y}+\frac{\partial \mathbf{F}_{z}^{\text{Visc}}}{\partial z}+S_{\text{MHD}},$$
(13)

where all flux and source terms are defined in detail in Appendix A. The state vector is defined as: $\mathbf{U} = (\rho, \rho u, \rho v, \rho w, B_x, B_y, B_z, E_{\text{tot}}, \epsilon_e)$.

2.2. The $\nabla \cdot \mathbf{B} = 0$ constraint

The presence of the $\nabla \cdot \mathbf{B} = 0$ constraint implies that the equations do not have a strictly hyperbolic character. It has been shown in [26] that even a small divergence in the magnetic fields can dramatically change the character of results from numerical simulations. In our work, we adopt an approach which was developed originally by Powell in [6]. The idea is to re-formulate the Jacobian matrix to include a "ninth-wave", i.e., the divergent mode that corresponds to velocity *u*. This way the degeneracy associated with the divergence-free condition is avoided while the rest of the eigenvalues of the Jacobian remain the same.

The primitive Jacobian matrix A_p for single-fluidlone-temperature equations has the form, in threedimensions

	u	ρ	0	0	0	0	0	0	ĺ
	0	и	0	0	$-\frac{B_x}{B_x}$	B_y	$\underline{B_z}$	1	
				0	$ ho B_v$	$\rho \\ B_r$	ρ	$\frac{ ho}{1}$	
	0	0	и	0	$-\frac{\gamma}{\rho}$	$\frac{n}{\rho}$	0	$\overline{\rho}$	
$A_p =$	0	0	0	и	$-\frac{B_z}{a}$	0	$\frac{B_x}{a}$	0	
	0	0	0	0	$\overset{\mathcal{P}}{0}$	0	$\stackrel{\rho}{0}$	0	
	0	B_y	$-B_x$	0	-v	и	0	0	
	0	B_z	0	$-B_x$	-w	0	и	0	
	0	γp	0	0	$-(\gamma - 1)\mathbf{u} \cdot \mathbf{B}$	0	0	и	

Considering that $p = p_i + p_e$, then the primitive Jacobian matrix A_p for single-fluid/two-temperature equations in three-dimensions can be described, i.e.

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

	Гu	ho	0	0	0	0	0	0	0	
	0	и	0	0	$-\frac{B_x}{\rho}$	$\frac{B_y}{\rho}$	$\frac{B_z}{\rho}$	$\frac{1}{\rho}$	0	
	0	0	и	0	$-\frac{B_y}{ ho}$	$\frac{B_x}{\rho}$	0	$\frac{1}{\rho}$	0	
$A_p =$	0	0	0	и	$-\frac{B_z}{ ho}$	0	$\frac{B_x}{\rho}$	0	0	
	0	0	0	0	0	0	0	0	0	
	0	B_y	$-B_x$	0	-v	и	0	0	0	
	0	B_z	0	$-B_x$	-w	0	и	0	0	
	0	γp	0	0	$-(\gamma - 1)\mathbf{u} \cdot \mathbf{B}$	0	0	и	0	
	0	γp_{e}	0	0	0	0	0	0	и	

To modify the governing equations so as to make A_p non-singular, using Powell's criteria presented in [6], A_p is modified to be A'_p

	Гu	ρ	0	0	0	0	0	0	0	
	0	и	0	0	0	$\frac{B_y}{\rho}$	$\frac{B_z}{\rho}$	$\frac{1}{\rho}$	0	
	0	0	и	0	0	$\frac{B_x}{\rho}$	0	$\frac{1}{\rho}$	0	
$A'_p =$	0	0	0	и	0	0	$\frac{B_x}{\rho}$	0	0	
	0	0	0	0	и	0	0	0	0	
	0	B_y	$-B_x$	0	0	и	0	0	0	
	0	B_z	0	$-B_x$	0	0	и	0	0	
	0	γp	0	0	0	0	0	и	0	
	0	γp_{e}	0	0	0	0	0	0	u	

This modification effectively corresponds to adding a source term proportional to $\nabla \cdot \mathbf{B}$,

$$\mathbf{S}_{\text{Powell}} = -(\nabla \cdot \mathbf{B})(0, B_x, B_y, B_z, u, v, w, \mathbf{u} \cdot \mathbf{B}, 0)^{\text{T}}$$

to the right-hand side of all evolution equations. We note that this source term does not shift the physical solution since $\nabla \cdot \mathbf{B}$ is imposed. However, some local accumulations may occur, especially at stagnation points for which v = 0. In these cases, it may be necessary to add the Helmholtz projection, see [26].

Next, we show how we need to modify these source terms in order to better maintain *pressure positivity*. In solving the MHD system, the pressure is a derived variable. Specifically, it is obtained by subtracting off the kinetic energy and magnetic energy from the total energy. However, in applications of micropropulsion, magneto-spheric physics and astrophysics, the pressure can be several orders of magnitude smaller than either the kinetic energy or the magnetic energy. Thus, small discretization errors in the total energy can produce situations where the pressure might become negative. This leads to an unacceptable physical situation. As long as the regions in front of a magneto-sonic shock have positive pressure, negative pressures would not be produced in magneto-sonic shocks.

Janhunen [27] has reported that the solution of the Riemann problem for Powell's equations for left- and right-states with positive fluid pressures may contain unphysical intermediate state with negative fluid pres-

$$\mathbf{S} = -(\nabla \cdot \mathbf{B})(0, 0, 0, 0, u, v, w, 0, 0)^{\mathsf{T}}$$

2.3. Implementation of the inviscid terms

We evaluate the inviscid fluxes and their derivatives in the interior of the elements and add correction terms (jumps) for the discontinuities in the flux between any two adjacent elements. In order to evaluate the Euler flux at an element interface, we use an one-dimensional Riemann solver to supply an upwinded flux there, see below. At a domain boundary, we provide far field conditions and treat the exterior boundary as the boundary of a "ghost" element. This way we can use the same Riemann solver at all element boundaries.

We linearize the one-dimensional flux $\mathbf{F}_{x}^{\text{Ideal}}$ in the normal direction to a shared element boundary using the *average* of the state vector at either side of the element boundary. That is, since $\mathbf{F}_{x}^{\text{Ideal}}$ is a nonlinear function of the state vector, we use the average state to form an approximation to the Jacobian of the flux vector A_c .

The Jacobian matrix for the flux vector of the evolution equations expressed in primitive variables is simpler than in the conserved form. Thus, we will perform the linearization about the primitive form and transform to the conserved form. The left and right eigenvectors of the primitive Jacobian matrix A_p , similar to the results shown in [6], are:

Entropy wave

$$\begin{aligned} \lambda_{\rm e} &= u, \\ l_{\rm e} &= \left(1, 0, 0, 0, 0, 0, 0, 0, -\frac{1}{a^2}, 0\right), \\ r_{\rm e} &= (1, 0, 0, 0, 0, 0, 0, 0, 0, 0)^{\rm t}. \end{aligned}$$

Alfven waves

$$\begin{split} \lambda_{\mathrm{a}} &= u \pm \frac{B_x}{\sqrt{\rho}}, \\ l_{\mathrm{a}} &= \frac{1}{\sqrt{2}} \left(0, 0, -\beta_z, \beta_y, 0, \pm \frac{\beta_z}{\sqrt{\rho}}, \mp \frac{\beta_y}{\sqrt{\rho}}, 0, 0 \right), \\ r_{\mathrm{a}} &= \frac{1}{\sqrt{2}} \left(0, 0, -\beta_z, \beta_y, 0, \pm \beta_z \sqrt{\rho}, \mp \beta_y \sqrt{\rho}, 0, 0 \right)^{\mathrm{T}} \end{split}$$

Fast waves

$$\begin{split} \lambda_{\rm f} &= u \pm c_{\rm f}, \\ l_{\rm f} &= \frac{1}{2a^2} \left(0, \pm \alpha_{\rm f} c_{\rm f}, \mp \alpha_{\rm s} c_{\rm s} \beta_x \beta_y, \mp \alpha_{\rm s} c_{\rm s} \beta_x \beta_z, 0, \frac{\alpha_{\rm s} \beta_y a}{\sqrt{\rho}}, \frac{\alpha_{\rm s} \beta_z a}{\sqrt{\rho}}, \frac{\alpha_{\rm f}}{\rho}, 0 \right), \\ r_{\rm f} &= \left(\rho \alpha_{\rm f}, \pm \alpha_{\rm f} c_{\rm f}, \mp \alpha_{\rm s} c_{\rm s} \beta_x \beta_y, \mp \alpha_{\rm s} c_{\rm s} \beta_x \beta_z, 0, \alpha_{\rm s} \beta_y a \sqrt{\rho}, \alpha_{\rm s} \beta_z a \sqrt{\rho}, \alpha_{\rm f} \gamma p, \alpha_{\rm f} \gamma p_{\rm e} \right)^{\rm t}. \end{split}$$

Slow waves 1

. . .

$$\lambda_{s} = u \pm c_{s},$$

$$l_{s} = \frac{1}{2a^{2}} \left(0, \pm \alpha_{s}c_{s}, \pm \alpha_{f}c_{f}\beta_{x}\beta_{y}, \pm \alpha_{f}c_{f}\beta_{x}\beta_{z}, 0, -\frac{\alpha_{f}\beta_{y}a}{\sqrt{\rho}}, -\frac{\alpha_{f}\beta_{z}}{\sqrt{\rho}}, \frac{\alpha_{s}}{\rho}, 0 \right),$$

$$r_{s} = \left(\rho\alpha_{s}, \pm \alpha_{s}c_{s}, \pm \alpha_{f}c_{f}\beta_{x}\beta_{y}, \pm \alpha_{f}c_{f}\beta_{x}\beta_{z}, 0, -\alpha_{f}\beta_{y}a\sqrt{\rho}, -\alpha_{f}\beta_{z}a\sqrt{\rho}, \alpha_{s}\gamma p, \alpha_{s}\gamma p_{e} \right)^{t}.$$

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

Compared to [6], we have an extra wave which corresponds to the electron energy:

$$l_{ee} = (0, 0, 0, 0, 0, 0, 0, 0, 1),$$

$$r_{ee} = \left(-\frac{\gamma p_e}{\rho}, 0, 0, 0, 0, 0, 0, 0, 1\right)^{t}.$$

Here

$$\begin{split} (a^*)^2 &= \frac{\gamma p + \mathbf{B} \cdot \mathbf{B}}{\rho}, \\ c_{\rm f}^2 &= \frac{1}{2} \left((a^*)^2 + \sqrt{(a^*)^4 - 4\frac{\gamma p B_x^2}{\rho^2}} \right), \quad c_{\rm s}^2 &= \frac{1}{2} \left((a^*)^2 - \sqrt{(a^*)^4 - 4\frac{\gamma p B_x^2}{\rho^2}} \right) \\ \alpha_{\rm f}^2 &= \frac{a^2 - c_{\rm s}^2}{c_{\rm f}^2 - c_{\rm s}^2}, \quad \alpha_{\rm s}^2 &= \frac{c_{\rm f}^2 - a^2}{c_{\rm f}^2 - c_{\rm s}^2}, \\ \beta_x &= \operatorname{sgn}(B_x), \quad \beta_y &= \frac{B_y}{\sqrt{B_y^2 + B_z^2}}, \quad \beta_z &= \frac{B_z}{\sqrt{B_y^2 + B_z^2}}. \end{split}$$

We can transform between the primitive variables W and conserved variables U with the following transform:

$$A_c = \frac{\partial \mathbf{U}}{\partial \mathbf{W}} A_p \frac{\partial \mathbf{W}}{\partial \mathbf{U}},$$

where

$$\mathbf{U} = (\rho, \rho u, \rho v, \rho w, B_x, B_y, B_z, E_{\text{tot}}, \rho \epsilon_e)$$

are the conserved variables, and

 $\mathbf{W} = (\rho, u, v, w, B_x, B_y, B_z, p, p_e)$

are the primitive variables. This gives

$$\frac{\partial \mathbf{U}}{\partial \mathbf{W}} = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ u & \rho & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ v & 0 & \rho & 0 & 0 & 0 & 0 & 0 & 0 \\ w & 0 & 0 & \rho & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 \\ \frac{uu}{2} & \rho u & \rho v & \rho w & B_x & B_y & B_z & \frac{1}{\gamma - 1} & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & \frac{1}{\gamma - 1} \end{bmatrix}$$

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

and

$$\frac{\partial \mathbf{W}}{\partial \mathbf{U}} = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ -\frac{u}{\rho} & \frac{1}{\rho} & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ -\frac{v}{\rho} & 0 & \frac{1}{\rho} & 0 & 0 & 0 & 0 & 0 & 0 \\ -\frac{w}{\rho} & 0 & 0 & \frac{1}{\rho} & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 \\ \frac{\bar{\gamma}}{2}\mathbf{u}\cdot\mathbf{u} & -\bar{\gamma}u & -\bar{\gamma}v & -\bar{\gamma}w & -\bar{\gamma}B_x & -\bar{\gamma}B_y & -\bar{\gamma}B_z & \bar{\gamma} & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & \bar{\gamma} \end{bmatrix}$$

where $\bar{\gamma} = \gamma - 1$.

We are now in a position to evaluate different fluxes at the element boundaries. The formulations of different fluxes we employ are:

(1) Flux employed in [28]

$$\widehat{\mathbf{F}}_{\text{upwind}}(\mathbf{U}_{\text{I}}, \mathbf{U}_{\text{E}}) = \frac{1}{2} \left[\mathbf{F}(\mathbf{U}_{\text{I}}) + \mathbf{F}(\mathbf{U}_{\text{E}}) - \frac{\partial \mathbf{U}}{\partial \mathbf{W}} \sum_{k=1}^{k=9} \alpha_{k} |\lambda_{k}| \mathbf{r}_{k} \right],$$

$$\alpha_{k} = \mathbf{l}_{k} \cdot \frac{\partial \mathbf{W}}{\partial \mathbf{U}} (\mathbf{U}_{\text{E}} - \mathbf{U}_{\text{I}}).$$
(14)

(2) Lax-Friedrichs flux

$$\widehat{\mathbf{F}}_{\text{Lax-Friedrichs}}(\mathbf{U}_{\text{I}}, \mathbf{U}_{\text{E}}) = \frac{1}{2} [\mathbf{F}(\mathbf{U}_{\text{I}}) + \mathbf{F}(\mathbf{U}_{\text{E}}) - D_{\max}(\mathbf{U}_{\text{E}} - \mathbf{U}_{\text{I}})],$$
(15)

where $D_{\max} = \max(|\lambda_k|)$ —maximal absolute value of the eigenvalues, 'I' denotes interior and 'E' denotes exterior of the element (see Fig. 1). Here, the \mathbf{l}_k and \mathbf{r}_k are the ordered *left* and *right* eigenvectors of the *primitive Jacobian matrix*. We have to apply the $\frac{\partial U}{\partial W}$ operator to the right eigenvectors to calculate the conserved flux. The λ_k 's are the wave speeds associated with the eigenvectors.

A new MHD flux introduced here is based on the MHD-HLLC flux presented in [25] that can preserve positivity and improve resolution, especially at contact interfaces.



Fig. 1. Interface conditions between two adjacent triangles.

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

(3) MHD-HLLC interface flux

$$\widehat{\mathbf{F}}_{\text{HLLC}}(\mathbf{U}_{\text{I}}, \mathbf{U}_{\text{E}}) = \begin{cases} \mathbf{F}_{\text{I}} & \text{if } S_{\text{I}} > 0, \\ \mathbf{F}_{\text{I}}^{*} & \text{if } S_{\text{I}} \leqslant 0 \leqslant S_{\text{M}}, \\ \mathbf{F}_{\text{E}}^{*} & \text{if } S_{\text{M}} \leqslant 0 \leqslant S_{\text{E}}, \\ \mathbf{F}_{\text{E}} & \text{if } S_{\text{E}} < 0, \end{cases}$$
(16)

where $\mathbf{F}_{\mathrm{I}}^*$ and $\mathbf{F}_{\mathrm{E}}^*$ are defined as

$$\begin{cases} \mathbf{F}_{\rm I}^* = \mathbf{F}^* - \frac{S_{\rm I}(S_{\rm E} - S_{\rm M})}{(S_{\rm E} - S_{\rm I})} \Delta \mathbf{U}^*, \\ \mathbf{F}_{\rm E}^* = \mathbf{F}^* + \frac{S_{\rm E}(S_{\rm M} - S_{\rm I})}{(S_{\rm E} - S_{\rm I})} \Delta \mathbf{U}^*. \end{cases}$$
(17)

with $\mathbf{F}^* = \frac{S_{\mathrm{E}}\mathbf{F}_{\mathrm{I}} - S_{\mathrm{I}}\mathbf{F}_{\mathrm{E}} + S_{\mathrm{I}}S_{\mathrm{E}}(\mathbf{U}_{\mathrm{E}} - \mathbf{U}_{\mathrm{I}})}{S_{\mathrm{E}} - S_{\mathrm{I}}}$ and $\Delta \mathbf{U}^* = \mathbf{U}_{\mathrm{E}}^* - \mathbf{U}_{\mathrm{I}}^*$. The wave-speeds S_{I} , S_{M} and S_{E} are defined as

$$\begin{cases} S_{\rm I} = \min[\lambda_{l}(\mathbf{U}_{\rm I}), \lambda_{l}(\mathbf{U}^{\rm Roe})], \\ S_{\rm M} = \frac{\rho_{\rm I} v_{n{\rm I}}(v_{n{\rm I}} - S_{\rm I}) - \rho_{\rm E} v_{n{\rm E}}(v_{n{\rm E}} - S_{\rm E}) + p_{\rm I} - p_{\rm E} + (|\mathbf{B}|_{\rm I}^{2} - |\mathbf{B}|_{\rm E}^{2})/2}{\rho_{\rm I}(v_{n{\rm I}} - S_{\rm I}) - \rho_{\rm E}(v_{n{\rm E}} - S_{\rm E})}, \\ S_{\rm E} = \min[\lambda_{m}(\mathbf{U}_{\rm E}), \lambda_{m}(\mathbf{U}^{\rm Roe})]. \end{cases}$$
(18)

Here $\lambda_l(\mathbf{U}^{\text{Roe}})$ and $\lambda_m(\mathbf{U}^{\text{Roe}})$ are the smallest and largest eigenvalues respectively, of the Roe-averaged matrix $A(\overline{\mathbf{U}})$, which satisfies the following property:

$$\mathbf{F}(\mathbf{U}_{\mathrm{E}}) - \mathbf{F}(\mathbf{U}_{\mathrm{I}}) = A(\overline{U})(\mathbf{U}_{\mathrm{E}} - \mathbf{U}_{\mathrm{I}}).$$
(19)

Correspondingly, $\lambda_l(\mathbf{U}_I)$ and $\lambda_m(\mathbf{U}_E)$ are the smallest and largest eigenvalues of the left and right states of the matrix A_p . The positivity of pressure and density using the MHD-HLLC flux has been demonstrated in [25]. Analytic results have shown that the flux resolves isolated contact discontinuities and fast waves accurately.

2.4. Implementation of the viscous terms

The viscous terms are evaluated in two steps. First, we obtain the spatial derivatives of the primitive variables using the discontinuous Galerkin approach. Then, we repeat the process for each of the viscous fluxes using these derivatives. If we employ Dirichlet boundary conditions for the momentum and energy variables, we set these terms explicitly after the fluxes have been evaluated and then project the result using the orthogonal basis. Here we use the average of the variables and fluxes at the interface. This approach leads to sub-optimal performance at low polynomial order p. However it does not make much difference at high polynomial order p. For more details, see [16].

3. Convergence and simulations

3.1. Convergence test

A simple test for the compressible MHD component of the algorithm we developed is to consider a steady irrotational magnetic field and zero velocity. The test was performed as an initial value problem, and the following exact solution:

10

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

$$\begin{split} \rho &= 1, \quad u = 0, \quad v = 0, \\ E &= 19.84 + \frac{\mathrm{e}^{(-2\pi y)}}{2}, \quad E_{\mathrm{e}} = 9.92, \\ B_x &= -\cos(\pi x)\mathrm{e}^{(-\pi y)}, \quad B_y = \sin(\pi x)\mathrm{e}^{(-\pi y)} \end{split}$$

was used as the *boundary conditions* and as the *initial condition*. This solution but without the part concerning E_e was derived by Priest [29]. The irrotational magnetic field implies that the Lorentz force is zero so the momentum equations are trivially satisfied. The magneto-viscous term is zero and the $\mathbf{v} \times \mathbf{B}$ term is also zero. Thus, the compressible single-fluid/two-temperature MHD equations are satisfied. Here, we take $m_i = 1836m_e$ [30] (m_i is the ion mass and m_e is the electron mass).



Fig. 2. Magneto-hydrostatic test case. Top-left: Spectral element hybrid mesh consisting of 38 triangular elements and 22 rectangular elements. Top-right: Magnetic streamlines of steady solution. Bottom: Dependence of steady state *log-error* of magnetic field B_x on expansion spectral order.

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

The domain and hybrid discretization we used are depicted in Fig. 2. We also show that the approximation error decreases exponentially with increasing expansion order for all the three forms of error considered in the L_{∞} , H_1 , and L_2 norms.



Fig. 3. 1D coplanar MHD Riemann problem. Solid line: MHD-HLLC flux; dashed line: Lax–Friedrichs flux. (a) ρ ; (b) p; (c) U_x ; (d) U_y ; (e) B_y .

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

3.2. Numerical tests for the MHD-HLLC interface flux

To verify our two-dimensional discontinuous Galerkin solver with the MHD-HLLC flux, we use the one-dimensional benchmark MHD shock-tube problem developed by Brio and Wu [31]. The one-dimensional Riemann problem is given for $x \in [-1, 1]$:

$$\mathbf{U}_{p} = (\rho, u_{x}, u_{y}, u_{z}, B_{y}, B_{z}, p) = \begin{cases} (1.000, 0, 0, 0, +1, 0, 1.0) & \text{for } x < 0, \\ (0.125, 0, 0, 0, -1, 0, 0.1) & \text{for } x > 0 \end{cases}$$
(20)

with $B_x = 0.75$ and $\gamma = 2$. The solution at t = 0.2 is shown in Fig. 3, which includes the left moving waves: the fast rarefaction wave, the intermediate shock attached by a slow rarefaction wave; and, the right moving waves: the contact discontinuity, a slow shock, and a fast rarefaction wave. The results are obtained by the two-dimensional discontinuous Galerkin solver on a mesh, consisting of 800 square elements. The solid line is the result using the MHD-HLLC flux, see Eq. (16), the dashed line is obtained using the Lax–Friedrichs flux, see Eq. (15). We can see the MHD-HLLC flux gives much sharper resolution, especially at the contact interface.



Fig. 4. High Mach number Riemann problem. Solid line: MHD-HLLC flux; dashed line: Lax–Friedrichs flux. (a) ρ ; (b) p; (c) B_{y} ; (d) U_{x} .

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

The second Riemann problem is given by

$$\mathbf{U}_{p} = (\rho, u_{x}, u_{y}, u_{z}, B_{y}, B_{z}, p) = \begin{cases} (1.000, 0, 0, 0, +1, 0, 1000.0) & \text{for } x < 0, \\ (0.125, 0, 0, 0, -1, 0, 0.1) & \text{for } x > 0 \end{cases}$$
(21)

with $B_x = 0$ and $\gamma = 2$. This problem is used to evaluate the code for high Mach number flow. If one regards the term $p + \frac{1}{2}|B|^2$ as the "hydrodynamic pressure", the system becomes a standard hydrodynamical Riemann problem. The computational domain is taken to be [-1, 1] with 400 square elements. The solution at t = 0.012 is shown in Fig. 4, which shows that the MHD-HLLC flux can resolve the high Mach number waves more accurately than the Lax–Friedrichs flux.

3.3. Flow past a cylinder

Next, we consider the problem of plasma flow with uniform free stream properties past a circular cylinder. As the mass and thermal properties for electrons and ions are quite different, at the final steady state, they will have quite different temperature distributions around the cylinder. Here we consider electrons and ions having the same temperature over the cylinder surface as the free stream temperature. To simplify our



Fig. 5. Top: Computational domain for plasma flow past a circular cylinder simulations at Mach number 0.7 and Reynolds number 100. Top-left: Entire domain. Top-right: Zoom around the cylinder. Bottom: Computational domain at Mach number 2 and Reynolds number 100. Bottom-left: Entire domain. Bottom-right: Zoom around the cylinder.

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

calculation, we set the atomic number of ions Z = 1 and $m_i = 1836m_e$. We perform simulations using unstructured meshes for all the subsonic and transonic cases; it is shown in Fig. 5 top, consisting of 490 triangular elements; the mesh used for the supersonic case is shown in Fig. 5 bottom, consisting of 1132 triangular elements.

h-refinement is employed around the shock, based on the following criterion (the gradient of density in the direction of the flow):

$$\nabla(\rho) \cdot \mathbf{V} > M,\tag{22}$$

where M is an adjustable parameter. The elements chosen are split into 4 smaller elements. Fig. 6 plots the computational domain showing the variable polynomial order; p-refinement is used away from the shock. To preserve solution monotonicity, we lower the spectral order around discontinuities appropriately. The polynomial order p is determined based on the area of the element but more sophisticated criteria can be used to find the optimal p. Due to the stability issue, first-order schemes or limiters have to be implemented around the shock. Since limiters will give more smearing results, we prefer to use low-order elements in conjunction with h-refinement close to the shock.



Fig. 6. Left: Variable order computational domain. Right: Variable order around the cylinder.



Fig. 7. Instantaneous non-dimensional temperature contours for flow past a circular cylinder at Mach number 0.7, Reynolds number 100 and cylinder wall temperature $T_i = T_c = 1.8367$. Left: Ion temperature contours. Right: Electron temperature contours.

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx



Fig. 8. Instantaneous non-dimensional temperature contours for flow past a circular cylinder at Mach number 2, Reynolds number 100 and cylinder wall temperature $T_i = T_c = 0.225$. Left: Ion temperature contours. Right: Electron temperature contours.



Fig. 9. Left: Time average contour lines of ion non-dimensional temperature (upper half plane) and electron non-dimensional temperature (lower half plane) at Mach number 0.7, Reynolds number 100 and cylinder wall temperature $T_i = T_e = 1.8367$ using the two-temperature model. Right: Time average contour lines of non-dimensional temperature using the one-temperature model.



Fig. 10. Left: Time average contour lines of ion non-dimensional temperature (upper half plane) and electron non-dimensional temperature (lower half plane) at Mach number 2, Reynolds number 100 and cylinder wall temperature $T_i = T_e = 0.225$ using the two-temperature model. Right: Time average contour lines of non-dimensional temperature using the one-temperature model.

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

In all the cases, we set Re = 100, $B_x = 0.1$ and $B_y = 0.0$ at the inflow, where B_x is the x-component of the magnetic field and B_y the y-component of the magnetic field. The two simulations were run with polynomial order p = 5 until the flow reaches a time-periodic or a steady state; p-refinement tests have shown only very small difference in the results.

In Figs. 7 and 8, we plot the ion and the electron temperature contours of the instantaneous field. A von Karman vortex street develops in the subsonic regime. However, the flow is steady at supersonic states as shown in Fig. 8. In particular, we have performed several simulations for the compressible flow, using the one-temperature model and the two-temperature model in order to present differences in the forces and frequency.

Specifically, in order to compare the one-temperature model with the two-temperature model, we integrate the flow field over one time period and obtain the time-averaged flow field. In Figs. 9 and 10, we plot the time-averaged contour lines of the ion and the electron temperature using the two-temperature model



Fig. 11. Location where profiles of temperature along the line shown aligned with the vertical axis are taken. Contours of ion temperature at Mach number 0.7 are shown in the background.



Fig. 12. Normal profiles of temperature non-dimensionalized by cylinder wall temperature at Mach number 0.7, Reynolds number 100 and cylinder wall temperature $T_i = T_e = 1.8367$. Solid line: ion temperature from two-temperature model; dotted line: electron temperature from two-temperature model; dashed line: temperature from one-temperature model.

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx



Fig. 13. Normal profiles of temperature non-dimensionalized by cylinder wall temperature at Mach number 2, Reynolds number 100 and cylinder wall temperature $T_i = T_e = 0.225$. Solid line: ion temperature from two-temperature model; dotted line: electron temperature from two-temperature model; dashed line: temperature from one-temperature model.

and the temperature contours using the one-temperature model. Also, we present temperature profiles along a line on top of the cylinder aligned with the vertical axis, as shown in Fig. 11, to compare the temperature distributions. In Figs. 12 and 13, we plot the normal profiles of temperature in one-temperature model and profiles of the electron and the ion temperature in two-temperature model starting at the top of the cylinder at Mach number 0.7 and 2. From these figures, we can see that the results of one-temperature model are quite different from the ones of the two-temperature model. In Figs. 12 and 13, electrons and ions have the same temperature on the cylinder surface and the electron temperature increases while the ion temperature decreases in the direction away from the cylinder surface. From Figs. 9 and 12, we can see that the ion temperature is larger than the electron temperature on the two sides of cylinder. Fig. 13 shows the electron temperature is smaller than the ion temperature on the two sides of cylinder. Fig. 13 shows the electron temperature profiles obtained from one temperature model is similar to the ion temperature rather than the electron temperature. The relation between two-temperature model and one-temperature model can be given as: $T = \frac{T_i+T_e}{2}$, if ions and electrons have the same number density $n_i = n_e$.

4. Summary

We have developed a discontinuous Galerkin solver to model two-temperature plasmas as part of a hierarchical modelling approach and a compromise between single-temperature and two-fluids models. We have demonstrated spectral convergence for an analytical problem and also demonstrated the robustness of the method in dealing with shocks without the use of flux limiters or artificial viscosity terms. The issue of preserving positivity is addressed by introducing a new interface flux, the MHD-HLLC flux.

Examination of contours of the divergence of the magnetic field in flow past a cylinder revealed some non-zero values in the region around the rear stagnation point. However, this was not growing in time and thus no numerical instabilities were induced even after long-time integration. This divergence field can be totally eliminated by occasional Helmholtz decomposition of the magnetic flux vector, see

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

[16,26]. We are currently developing a discontinuous Galerkin method for two-fluids plasmas and we will report those results in a future publication.

Acknowledgements

This work was supported by the computational mathematics program of AFOSR. We would like to acknowledge helpful discussions with Prof. N. Gatsonis regarding this work. Computations were performed at NPACI's SP3.

Appendix A. Detailed flux terms in single-fluid/two-temperature equations

We present here in detail the flux terms involved in the governing equations of the single-fluid/two-temperature plasma. Many of the parameters used are listed in Table 1.

$$\begin{split} \mathbf{F}_{x}^{\text{ideal}} &= \left(\rho u, \rho u^{2} - B_{x}^{2} + \bar{p}, \rho uv - B_{x}B_{y}, \rho uw - B_{x}B_{z}, 0, uB_{y} - vB_{x}, uB_{z} - wB_{x}, \\ &(E + \bar{p})u - (\mathbf{v} \cdot \mathbf{B})B_{x}, (\epsilon_{e} + p_{e})u)^{\mathsf{T}}, \\ \mathbf{F}_{y}^{\text{ideal}} &= \left(\rho v, \rho vu - B_{y}B_{x}, \rho v^{2} - B_{y}^{2} + \bar{p}, \rho vw - B_{y}B_{z}, vB_{x} - uB_{y}, 0, vB_{z} - wB_{y}, \\ &(E + \bar{p})v - (\mathbf{v} \cdot \mathbf{B})B_{y}, (\epsilon_{e} + p_{e})v)^{\mathsf{T}}, \\ \mathbf{F}_{z}^{\text{ideal}} &= \left(\rho w, \rho vu - B_{z}B_{x}, \rho wv - B_{z}B_{y}, \rho w^{2} - B_{z}^{2} + \bar{p}, wB_{x} - uB_{z}, wB_{y} - vB_{z}, \\ &0, (E + \bar{p})w - (v \cdot \mathbf{B})B_{z}, (\epsilon_{e} + p_{e})w)^{\mathsf{T}}, \\ \mathbf{F}_{x}^{\text{vise}} &= \left(0, \frac{2}{S_{vi}} \left(\frac{\partial u}{\partial x} - \frac{1}{3}\nabla \cdot \mathbf{v}\right), \frac{1}{S_{vi}} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}\right), \frac{1}{S_{vi}} \left(\frac{\partial u}{\partial z} + \frac{\partial w}{\partial y}\right), 0, -\frac{1}{S_{r}} \left(\frac{\partial B_{y}}{\partial x} - \frac{\partial B_{x}}{\partial y}\right), \\ &-\frac{1}{S_{r}} \left(\frac{\partial B_{z}}{\partial x} - \frac{\partial B_{x}}{\partial z}\right), \frac{1}{S_{vi}} \left(-\frac{2}{3}(\nabla \cdot \mathbf{v})u + \mathbf{v} \cdot \nabla u + \frac{1}{2}\frac{\partial \mathbf{v}^{2}}{\partial x} + \frac{1}{Pr_{i}}\frac{\partial T_{i}}{\partial x} + \frac{1}{Pr_{c}}\frac{\partial T_{e}}{\partial x}\right) \\ &+\frac{1}{S_{r}} \left(\frac{1}{2}\frac{\partial (|\mathbf{B}|^{2})}{\partial x} - \mathbf{B} \cdot \nabla B_{x}\right), \frac{1}{S_{ve}Pr_{e}}\frac{\partial T_{e}}{\partial x}\right)^{\mathsf{T}}, \\ \mathbf{F}_{y}^{\text{vise}} &= \left(0, \frac{1}{S_{vi}} \left(\frac{\partial v}{\partial x} + \frac{\partial u}{\partial y}\right), \frac{2}{S_{vi}} \left(\frac{\partial v}{\partial y} - \frac{1}{3}\nabla \cdot \mathbf{v}\right), \frac{1}{S_{vi}} \left(\frac{\partial v}{\partial z} + \frac{\partial w}{\partial y}\right), -\frac{1}{S_{r}} \left(\frac{\partial B_{x}}{\partial y} - \frac{\partial B_{y}}{\partial x}\right), 0, \\ &-\frac{1}{S_{r}} \left(\frac{\partial B_{z}}{\partial y} - \frac{\partial B_{y}}{\partial z}\right), \frac{1}{S_{vi}} \left(-\frac{2}{3}(\nabla \cdot \mathbf{v})v + \mathbf{v} \cdot \nabla v + \frac{1}{2}\frac{\partial v^{2}}{\partial y} + \frac{1}{Pr_{i}}\frac{\partial T_{i}}{\partial y} + \frac{1}{Pr_{e}}\frac{\partial T_{e}}{\partial y}\right) \\ &+\frac{1}{S_{r}} \left(\frac{1}{2}\frac{\partial (|\mathbf{B}|^{2})}{\partial y} - \mathbf{B} \cdot \nabla B_{y}\right), \frac{1}{S_{ve}Pr_{e}}\frac{\partial T_{e}}{\partial y}\right)^{\mathsf{T}}, \\ \mathbf{F}_{z}^{\text{vise}} &= \left(0, \frac{1}{S_{vi}} \left(\frac{\partial w}{\partial x} + \frac{\partial u}{\partial z}\right), \frac{1}{S_{vi}} \left(\frac{\partial w}{\partial y} + \frac{\partial v}{\partial z}\right), \frac{2}{S_{vi}} \left(\frac{\partial w}{\partial z} - \frac{1}{3}\nabla \cdot \mathbf{v}\right), -\frac{1}{S_{r}} \left(\frac{\partial B_{x}}{\partial z} - \frac{\partial B_{z}}{\partial x}\right), \\ &-\frac{1}{S_{r}} \left(\frac{\partial B_{y}}{\partial z} - \frac{\partial B_{z}}{\partial y}\right), 0, \frac{1}{S_{vi}} \left(\frac{\partial w}{\partial y} + \frac{\partial v}{\partial z}\right), \frac{2}{S_{vi}} \left(\frac{\partial w}{\partial z} - \frac{1}{3}\nabla \cdot \mathbf{v}\right), -\frac{1}{S_{r}} \left(\frac{\partial B_{x}}{\partial z} - \frac{\partial B_$$

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

Appendix B. From two-fluid equations to one-fluid two-temperature equations

The magneto-hydrodynamic model treats the plasma as a single fluid. In the following, we derive the single-fluid/two-temperature plasma equations from the two-fluids plasma equations. We denote the electron and ion masses by *m* and *M* respectively. n_i and n_e are the ion and electron number density. j_i , j_e , T_i , T_e , p_i , p_e and E_i , E_e are the current density, temperature pressure and hydrodynamic energy density of ion and electron respectively. E_{tot} and ϵ_e are the total energy density and electron internal energy density.

We define:

$$M = 1836n, \quad n_{\rm i} = n_{\rm e} = n, \tag{23}$$

$$\rho = n_i M + n_e m = n(M+m) \approx nM, \tag{24}$$

$$i = n_i V. \tag{25}$$

$$\mathbf{j}_{\mathbf{i}} = n_{\mathbf{e}} \mathbf{v}_{\mathbf{i}},$$

$$\mathbf{j}_{\mathbf{e}} = -n_{\mathbf{e}} \mathbf{v}_{\mathbf{e}},$$
(26)

$$\mathbf{j} = n_{\mathrm{e}}(\mathbf{v}_{\mathrm{i}} - \mathbf{v}_{\mathrm{e}}) = n_{\mathrm{e}}(\mathbf{j}_{\mathrm{i}} + \mathbf{j}_{\mathrm{e}}), \tag{27}$$

$$p = p_{i} + p_{e},$$

$$T = \frac{T_{i} + T_{e}}{2},$$
(28)
(29)

$$\mathbf{v} = \frac{n(M\mathbf{v}_{i} + m\mathbf{v}_{e})}{n} = \frac{M\mathbf{v}_{i} + m\mathbf{v}_{e}}{M + m} \approx \mathbf{v}_{i} + \frac{m}{M}\mathbf{v}_{e} \approx \mathbf{v}_{i},$$
(30)

$$\mathbf{v}_{i} \approx \mathbf{v} + \frac{m\mathbf{j}}{Mn_{e}} \approx \mathbf{v},\tag{31}$$

$$\mathbf{v}_{\mathrm{e}} \approx \mathbf{v} - \frac{\mathbf{j}}{n_{\mathrm{e}}} \approx -\frac{\mathbf{j}}{n_{\mathrm{e}}},$$
(32)

$$E_{i} = \frac{nkT_{i}}{\gamma - 1} + \frac{1}{2}nM\mathbf{v}_{i}^{2},$$
(33)

$$E_{\rm e} = \frac{nkT_{\rm e}}{\gamma - 1} + \frac{1}{2}nm\mathbf{v}_{\rm e}^2,\tag{34}$$

$$E_{\text{tot}} = E_{\text{i}} + E_{\text{e}} + \frac{1}{2}\mathbf{B}^2,\tag{35}$$

$$\epsilon_{\rm e} = \frac{nkT_{\rm e}}{\gamma - 1},\tag{36}$$

where $\gamma = 5/3$ and k is the Boltzmann constant.

We make following assumptions:

1. Quasineutral approximation

$$n = n_{\rm i} = n_{\rm e}.\tag{37}$$

- 2. *High collisionality*. We assume both the electrons and ions are collision dominated. The collisions rapidly randomize the distribution function giving rise to an isotropic pressure.
- 3. Small Larmor radius. Generalized Ohm's law:

$$\mathbf{E} = \eta \mathbf{j} + \mathbf{v} \times \mathbf{B} + \frac{\mathbf{j} \times \mathbf{B} - \nabla p_{\mathrm{e}}}{n_{\mathrm{e}}}.$$
(38)

With the 'small Larmor radius' approximation, the generalized Ohm's law is simplified as

$$\mathbf{E} = \eta \mathbf{j} - \mathbf{v} \times \mathbf{B}. \tag{39}$$

20

4. Electrons move much faster than ions.

$$|\mathbf{v}_i| \ll |\mathbf{v}_e|.$$

B.1. Mass conservation

In two-fluids plasma mass conservation equations, we have

$$\frac{\partial n_{i}}{\partial t} + \nabla \cdot (n_{i}\mathbf{v}_{i}) = 0, \tag{41}$$

$$\frac{\partial n_{e}}{\partial t} + \nabla \cdot (n_{e}\mathbf{v}_{e}) = 0. \tag{42}$$

By multiplying the ion and electron masses M and m, respectively, and adding the above two equations together, we produce the 'single-fluid mass conservation equation'

$$\frac{\partial n(M+m)}{\partial t} + \nabla \cdot [n(M\mathbf{v}_{i} + m\mathbf{v}_{e})] = \frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0.$$
(43)

B.2. Momentum conservation

In two-fluids plasma, the momentum conservation equations are:

$$\frac{\partial (nM\mathbf{v}_{i})}{\partial t} + \nabla \cdot (nM\mathbf{v}_{i}\mathbf{v}_{i}^{t}\mathbf{I}) = -\nabla p_{i} + n_{e}(\mathbf{E} + \mathbf{v}_{i} \times \mathbf{B}) + \nabla \cdot (\mu_{i}\tau_{i}),$$
(44)

$$\frac{\partial(nm\mathbf{v}_{e})}{\partial t} + \nabla \cdot (nm\mathbf{v}_{e}\mathbf{v}_{e}^{t}\mathbf{I}) = -\nabla p_{e} - n_{e}(\mathbf{E} + \mathbf{v}_{e} \times \mathbf{B}) + \nabla \cdot (\mu_{e}\tau_{e}),$$
(45)

where $\boldsymbol{\tau}_{i} = (\partial_{j} \mathbf{v}_{ii} + \partial_{i} \mathbf{v}_{ij}) - \frac{2}{3} \nabla \cdot \mathbf{v}_{i} \delta_{ij}$ and $\boldsymbol{\tau}_{e} = (\partial_{j} \mathbf{v}_{ei} + \partial_{i} \mathbf{v}_{ej}) - \frac{2}{3} \nabla \cdot \mathbf{v}_{e} \delta_{ij}$. By adding the above two equations together, we have

$$\frac{\partial(\rho \mathbf{v})}{\partial t} + \nabla \cdot (nM \mathbf{v}_{i} \mathbf{v}_{i}^{t} \mathbf{I} + nm \mathbf{v}_{e} \mathbf{v}_{e}^{t} \mathbf{I})$$

$$\approx \frac{\partial(\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v} \mathbf{v}^{t} \mathbf{I})$$

$$= -\nabla(p_{i} + p_{e}) + n_{e}(\mathbf{v}_{i} - \mathbf{v}_{e}) \times \mathbf{B} + \nabla \cdot (\mu_{i} \tau_{i} + \mu_{e} \tau_{e})$$

$$= -\nabla p + \mathbf{j} \times \mathbf{B} + \nabla \cdot (\mu_{i} \tau_{i} + \mu_{e} \tau_{e}),$$
(46)

where we define $p = p_i + p_e$ and $\mathbf{j} = ne(\mathbf{v}_i - \mathbf{v}_e)$. Since $\mathbf{j} \times \mathbf{B} = \frac{1}{\mu_0} (\nabla \times \mathbf{B}) \times \mathbf{B} = -\frac{1}{\mu_0} \nabla \cdot \left(-\mathbf{B}\mathbf{B}^t + \frac{1}{2} |\mathbf{B}|^2 \mathbf{I} \right)$, the combined 'single-fluid momentum conservation equation' is obtained as follows:

$$\frac{\partial(\rho \mathbf{v})}{\partial t} = -\nabla \cdot \left(\rho \mathbf{v} \mathbf{v}^{t} - \frac{\mathbf{B} \mathbf{B}^{t}}{\mu_{0}} + \left(p + \frac{1}{2\mu_{0}} |\mathbf{B}|^{2}\right) \mathbf{I} - \mu_{i} \tau_{i} - \mu_{e} \tau_{e}\right).$$
(47)

After non-dimensionalization, we have

$$\frac{\partial(\rho \mathbf{v})}{\partial t} = -\nabla \cdot \left(\rho \mathbf{v} \mathbf{v}^{t} - \mathbf{B} \mathbf{B}^{t} + \left(p + \frac{1}{2} |\mathbf{B}|^{2}\right) \mathbf{I} - \frac{1}{S_{vi}} \boldsymbol{\tau}_{i} - \frac{1}{S_{ve}} \boldsymbol{\tau}_{e}\right),\tag{48}$$

where $S_{vi} = \frac{\rho_0 V_A L_0}{\mu_i}$ and $S_{ve} = \frac{\rho_0 V_A L_0}{\mu_e}$.

(40)

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

B.3. Magnetic field

$$\frac{\partial \mathbf{B}}{\partial t} = -\nabla \times \mathbf{E},\tag{49}$$

where $\mathbf{E} = \frac{1}{\sigma \mu_0} \nabla \times \mathbf{B} - \mathbf{v} \times \mathbf{B}$.

$$\frac{\partial \mathbf{B}}{\partial t} = -\nabla \times \left(\mathbf{B} \times \mathbf{v} + \frac{1}{\sigma \mu_0} \nabla \times \mathbf{B} \right),\tag{50}$$

where $\mathbf{v} = \frac{M\mathbf{v}_i + m\mathbf{v}_e}{M + m}$. After non-dimensionalization, we have

$$\frac{\partial \mathbf{B}}{\partial t} = -\nabla \times \left(\mathbf{B} \times \mathbf{v} + \frac{1}{S_r} \nabla \times \mathbf{B} \right),\tag{51}$$

where $S_r = \mu_0 V_A L_0 / \eta$.

B.4. Energy conservation

In two-fluids plasma energy conservation equations, we have

$$\frac{\partial E_{i}}{\partial t} + \nabla \cdot \left[(E_{i} + p_{i})\mathbf{v}_{i} \right] = \mathbf{j}_{i} \cdot \mathbf{E} + \nabla \cdot (\mu_{i}\mathbf{v}_{i} \cdot \boldsymbol{\tau}_{i} + \kappa_{i}\nabla T_{i}),$$

$$\frac{\partial E_{i}}{\partial E_{i}}$$
(52)

$$\frac{\partial E_{e}}{\partial t} + \nabla \cdot \left[(E_{e} + p_{e}) \mathbf{v}_{e} \right] = \mathbf{j}_{e} \cdot \mathbf{E} + \nabla \cdot (\mu_{e} \mathbf{v}_{e} \cdot \boldsymbol{\tau}_{e} + \kappa_{e} \nabla T_{e}).$$
(53)

By substituting $v_i\approx v+\frac{mj}{Mn_e}$ and $v_e\approx v-\frac{j}{n_e}$ into the above two equations, we have

$$\frac{\partial E_{i}}{\partial t} + \nabla \cdot \left[(E_{i} + p_{i})\mathbf{v} \right] + \frac{m}{M\mu_{0}ne} \nabla \cdot \left[(E_{i} + p_{i})(\nabla \times \mathbf{B}) \right] = \mathbf{j}_{i} \cdot \mathbf{E} + \nabla \cdot (\mu_{i}\mathbf{v}_{i} \cdot \boldsymbol{\tau}_{i} + \kappa_{i}\nabla T_{i}),$$
(54)

$$\frac{\partial E_{\rm e}}{\partial t} + \nabla \cdot \left[(E_{\rm e} + p_{\rm e}) \mathbf{v} \right] - \frac{1}{\mu_0 n e} \nabla \cdot \left[(E_{\rm e} + p_{\rm e}) (\nabla \times \mathbf{B}) \right] = \mathbf{j}_{\rm e} \cdot \mathbf{E} + \nabla \cdot (\mu_{\rm e} \mathbf{v}_{\rm e} \cdot \tau_{\rm e} + \kappa_{\rm e} \nabla T_{\rm e}). \tag{55}$$

Adding the above two equations and considering in two-dimensional space, $\nabla \cdot (\nabla \times \mathbf{B}) = 0$, the above equation can be simplified as

$$\frac{\partial E_{\text{tot}}}{\partial t} + \nabla \cdot \left[(E_{\text{tot}} + p) \mathbf{v} \right] = \mathbf{j} \cdot \mathbf{E} + \nabla \cdot (\mu_{\text{i}} \mathbf{v}_{\text{i}} \cdot \boldsymbol{\tau}_{\text{i}} + \mu_{\text{e}} \mathbf{v}_{\text{e}} \cdot \boldsymbol{\tau}_{\text{e}} + k_{\text{i}} \nabla T_{\text{i}} + \kappa_{\text{e}} \nabla T_{\text{e}}).$$
(56)

Using the magnetic field equation obtained above, we have

$$\frac{1}{2}\frac{\partial \mathbf{B}^2}{\partial t} = -\mathbf{B}\cdot\nabla\times(\mathbf{B}\times\mathbf{v}+\eta\nabla\times\mathbf{B}) = -\nabla\cdot\left[\left(\frac{1}{2}|\mathbf{B}|^2I - \mathbf{B}\mathbf{B}^t\right)\cdot\mathbf{v}+\eta\mathbf{j}\times\mathbf{B}\right] - \mathbf{j}\cdot\mathbf{E}.$$
(57)

Adding the above two equations, we have

$$\frac{\partial E_{\text{tot}}}{\partial t} + \nabla \cdot \left[(E_{\text{tot}} + p) \mathbf{v} \right] = \mathbf{j} \cdot \mathbf{E} + \nabla \cdot (\mu_{i} \mathbf{v}_{i} \cdot \boldsymbol{\tau}_{i} + \mu_{e} \mathbf{v}_{e} \cdot \boldsymbol{\tau}_{e} + k_{i} \nabla T_{i} + k_{e} \nabla T_{e}) - \nabla \cdot \left[\left(\frac{1}{2} |\mathbf{B}|^{2} I - \mathbf{B} \mathbf{B}^{t} \right) \cdot \mathbf{v} + \eta \mathbf{j} \times \mathbf{B} \right] - \mathbf{j} \cdot \mathbf{E}.$$
(58)

Since

$$(\nabla \times \mathbf{B}) \times \mathbf{B} = \mathbf{B} \cdot \nabla \mathbf{B} - \nabla \frac{|\mathbf{B}|^2}{2},$$
(59)

22

we have the combined energy conservation equation:

$$\frac{\partial E_{\text{tot}}}{\partial t} = -\nabla \cdot \left[(E_{\text{tot}} + p) \mathbf{v} + \left(\frac{1}{2} |\mathbf{B}|^2 I - \mathbf{B} \mathbf{B}^{\text{t}} \right) \cdot \mathbf{v} - \mu_{\text{i}} \mathbf{v} \cdot \boldsymbol{\tau}_{\text{i}} - \mu_{\text{e}} \mathbf{v} \cdot \boldsymbol{\tau}_{\text{e}} - k_{\text{i}} \nabla T_{\text{i}} - k_{\text{e}} \nabla T_{\text{e}} + \eta \left(\mathbf{B} \cdot \nabla \mathbf{B} - \nabla \frac{|\mathbf{B}|^2}{2} \right) \right].$$
(60)

After non-dimensionalization, we have

$$\frac{\partial E_{\text{tot}}}{\partial t} = -\nabla \cdot \left[(E_{\text{tot}} + p) \mathbf{v} + \left(\frac{1}{2} |\mathbf{B}|^2 I - \mathbf{B} \mathbf{B}^{\text{t}} \right) \cdot \mathbf{v} - \frac{\gamma}{S_{\text{vi}} P r_{\text{i}}} \nabla T_{\text{i}} - \frac{\gamma}{S_{\text{ve}} P r_{\text{e}}} \nabla T_{\text{e}} - \frac{1}{S_{\text{vi}}} \mathbf{v} \cdot \boldsymbol{\tau}_{\text{i}} - \frac{1}{S_{\text{vi}}} \mathbf{v} \cdot \boldsymbol{\tau}_{\text{e}} + \frac{1}{S_{r}} \left(\mathbf{B} \cdot \nabla \mathbf{B} - \nabla \frac{|\mathbf{B}|^2}{2} \right) \right],$$
(61)

where $Pr_i = \frac{C_p \mu_i}{\kappa_i}$ and $Pr_e = \frac{C_p \mu_e}{\kappa_e}$.

B.5. Electron energy conservation

The electron energy conservation equation is

$$\frac{\partial \epsilon_{\rm e}}{\partial t} + \nabla \cdot \left[(\epsilon_{\rm e} + p_{\rm e}) \mathbf{v}_{\rm e} \right] = \nabla \cdot (\kappa_{\rm e} \nabla T_{\rm e}) + \mu_{\rm e} \boldsymbol{\tau}_{\rm e} : \nabla \mathbf{v}_{\rm e} + \mathbf{v}_{\rm e} \cdot \nabla p_{\rm e} + \eta \mathbf{j} \cdot \mathbf{j}.$$
(62)

By substituting $\mathbf{v}_{e} \approx \mathbf{v} - \frac{\mathbf{j}}{n_{e}}$ into above equation, we have

$$\frac{\partial \epsilon_{\rm e}}{\partial t} + \nabla \cdot \left[(\epsilon_{\rm e} + p_{\rm e}) \mathbf{v} \right] - \frac{1}{\mu_0 n e} \nabla \cdot \left[(\epsilon_{\rm e} + p_{\rm e}) (\nabla \times \mathbf{B}) \right] = \nabla \cdot (\kappa_{\rm e} \nabla T_{\rm e}) + \mu_{\rm e} \tau_{\rm e} : \nabla \mathbf{v}_{\rm e} + \mathbf{v} \cdot \nabla p_{\rm e} - \frac{(\nabla \times \mathbf{B})}{\mu_0 n_{\rm e}} \cdot \nabla p_{\rm e} + \eta \mathbf{j} \cdot \mathbf{j}.$$

$$(63)$$

Considering the two-dimensional space, $\nabla \cdot (\nabla \times \mathbf{B}) = 0$ and $\frac{(\nabla \times \mathbf{B})}{\mu_0 n_e} \cdot \nabla p_e = 0$, above equation can be simplified:

$$\frac{\partial \epsilon_{\rm e}}{\partial t} + \nabla \cdot \left[(\epsilon_{\rm e} + p_{\rm e}) \mathbf{v} \right] = \nabla \cdot (\kappa_{\rm e} \nabla T_{\rm e}) + \mu_{\rm e} \boldsymbol{\tau}_{\rm e} : \nabla \mathbf{v} + \mathbf{v} \cdot \nabla p_{\rm e} + \eta \mathbf{j} \cdot \mathbf{j}.$$
(64)

After non-dimensionalization, we have

$$\frac{\partial \epsilon_{\rm e}}{\partial t} = -\nabla \cdot \left[(\epsilon_{\rm e} + p_{\rm e}) \mathbf{v} - \frac{\gamma}{S_{\rm ve} P r_{\rm e}} \nabla T_{\rm e} \right] + \mathbf{v} \cdot \nabla p_{\rm e} + \frac{1}{S_{\rm ve}} \tau_{\rm e} : \nabla \mathbf{v} + \frac{1}{S_r} (\nabla \times \mathbf{B}) \cdot (\nabla \times \mathbf{B}).$$
(65)

References

- N.A. Gatsonis, X. Yin, Hybrid (particle/fluid) modeling of pulsed plasma thruster plumes, J. Spacecraft Rockets 17 (2001) 945– 958.
- [2] N.A. Gatsonis, R.A. Nanson, G.J. Le Beau, Simulations of cold-gas nozzle and plume flows and flight data comparisons, J. Spacecraft Rockets 37 (2000) 39–48.
- [3] W. Dai, P.R. Woodward, A simple Riemann solver and high-order Godunov schemes for hyperbolic systems of conservation laws, J. Comput. Phys. 121 (1995) 51.
- [4] A.L. Zachary, A. Malagoli, P. Colella, A higher-order Godunov method for multidimensional ideal magnetohydrodynamics, SIAM J. Sci. Stat. Comput. 15 (1994) 15.
- [5] R.E. Peterkin, M.H. Frese, C.R. Sovinec, Transport magnetic flux in an arbitrary coordinate ALE code, J. Comput. Phys. 140 (1998) 148.

24

G. Lin, G.E. Karniadakis / Comput. Methods Appl. Mech. Engrg. xxx (2005) xxx-xxx

- [6] K.G. Powell, An approximate Riemann solver for magnetohydrodynamics (that works in more than one dimension), ICASE Report No. 94-24, 1994.
- [7] O.S. Jones, U. Shumlak, D.S. Eberhardt, An implicit scheme for nonideal magnetohydrodynamics, J. Comput. Phys. 130 (1997) 231.
- [8] P. Colella, M. Dorr, D.D. Wake, A conservative finite difference method for the numerical solution of plasma fluid equations, Technical Report UCRL-JC-129912, 1998.
- [9] P.J. Turchi, P.G. Mikellides, Modeling of ablation-fed pulsed plasma thrusters, in: 31st AIAA/ASME/SAE/ASEE Joint Propulsion Conference and Exhibit, July 10–12 1995.
- [10] U. Shumlak, B. Udrea, An approximate Riemann solver for MHD computations on parallel architectures, in: AIAA 2001-2591, 15th AIAA Computational Fluid Dynamics Conference, Anaheim, CA, 11–14 June 2001.
- [11] J.P. Freidber, Ideal magnetohydrodynamic theory of magnetic fusion systems, Rev. Mod. Phys. 54 (3) (1982) 801.
- [12] U. Shumlak, J. Loverich, Approximate Riemann solver for the two-fluid plasma model, J. Comput. Phys. 187 (2003) 620.
- [13] R.T. Brown, M. Mitchner, Measurements of a nonequilibrium plasma boundary layer, Phys. Fluids 14 (5) (1971) 933.
- [14] J.R. Roth, D.M. Sherman, Boundary layer flow control with a one atmosphere uniform glow discharge surface plasma, in: AIAA 98-0328, 36th Aerospace Sciences Meeting & Exhibit, Reno, NV, 12–15 January 1998.
- [15] S.P. Wilkinson, Investigation of an oscillating surface plasma for turbulent drag reduction, in: AIAA 2003-1023, 41st Aerospace Sciences Meeting & Exhibit, Reno, NV, 6–9 January 2003.
- [16] T.C. Warburton, G.E. Karniadakis, A discontinuous Galerkin method for the viscous MHD equations, J. Comput. Phys. 152 (1999) 933.
- [17] R.B. Dahlburg, J.M. Picone, Evolution of the Orszag–Tang vortex system in a compressible medium. I. Initial average subsonic flow, Phys. Fluids B 1 (11) (1989).
- [18] H.O. Kreiss, J. Olinger, Methods for the Approximate Solution of Time-Dependent Problems, GARP Publ. Ser., vol. 10, GARP, Geneva, 1973.
- [19] D.V. Gaitonde, High-order solution procedure for three-dimensional nonideal magnetogasdynamics, AIAA J. 39 (11) (2001) 2111.
- [20] D.V. Gaitonde, J. Poggie, Elements of a numerical procedure for 3-D MGD flow control analysis, in: AIAA 2002-0198, 40th Aerospace Sciences Meeting & Exhibit, Reno, NV, 14–17 January 2002.
- [21] A.H. Glasser, X.Z. Tang, Progress in the development of the SEL macroscopic modeling code, in: Sherwood Fusion Theory Conference, Corpu Christi, TX, 28–30 April 2003.
- [22] B. Cockburn, G.E. Karniadakis, C.-W. Shu, The development of discontinuous Galerkin methods, in: Discontinuous Galerkin Methods: Theory, Computation and Applications, Springer, 2000.
- [23] C.R. Evans, J.F. Hawley, Simulation of magnetohydrodynamic flows: a constrained transport method, Astro. Phys. J. 332 (1988) 659–677.
- [24] G.E. Karniadakis, S.J. Sherwin, Spectral/hp Element Methods for CFD, Oxford University Press, 1999.
- [25] K.F. Gurski, An HLLC-type approximate Riemann solver for ideal magnetohydrodynamics, SIAM J. Sci. Comput. 25 (2004) 2165–2187.
- [26] J.U. Brackbill, D.C. Barnes, The effect of nonzero ∇ · b on the numerical solution of the magnetohydrodynamic equations, J. Comput. Phys. 35 (1980) 426.
- [27] P. Janhunen, A positive conservative method for magnetohydrodynamics based on HLL and Roe methods, J. Comput. Phys. 160 (2000) 649–661.
- [28] K.G. Powell, P.L. Roe, R.S. Myong, T.I. Gombosi, D.L. DeZeeuw, An upwind scheme for magnetohydrodynamics, in: AIAA 12th Computational Dynamics Conference, San Diego, CA, AIAA-95-1704, 1995, p. 661.
- [29] E. Priest, Solar Magnetohydrodynamics, D. Reidel Publishing Company, 1982.
- [30] M. Mitchner, C.H. Kruger Jr., Partially Ionized Gases, John Wiley & Sons, Inc., 1973.
- [31] M. Brio, C.C. Wu, An upwind differencing scheme for the equations of ideal magnetohydrodynamics, J. Comput. Phys. 75 (1988) 400.